

# **COE CST Third Annual Technical Meeting: Ultrahigh Temperature Composites for Thermal Protection Systems (TPS)**

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***October 29, 2013***



# Overview

- Team Members
- Purpose of Task
- Research Methodology
- Results or Schedule & Milestones
- Next Steps
- Contact Information

# Team Members

## Principle Investigators

- **Jan Gou** - Composites design and manufacturing, composites mechanics
- **Jay Kapat** - Heat transfer, film cooling, aerodynamics testing
- **Ali Gordon** - Thermo-mechanical testing and modeling

## Graduate Students

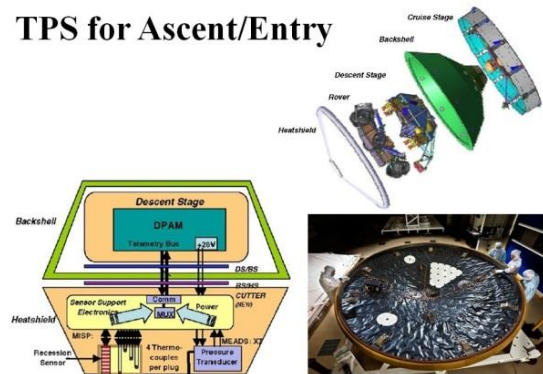
- **James McKee, Hongjiang Yang**: Composites TPS design & manufacturing
- **Donovan Lui**: Ablation testing
- **Cassandra Carpenter**: Aerothermal modeling



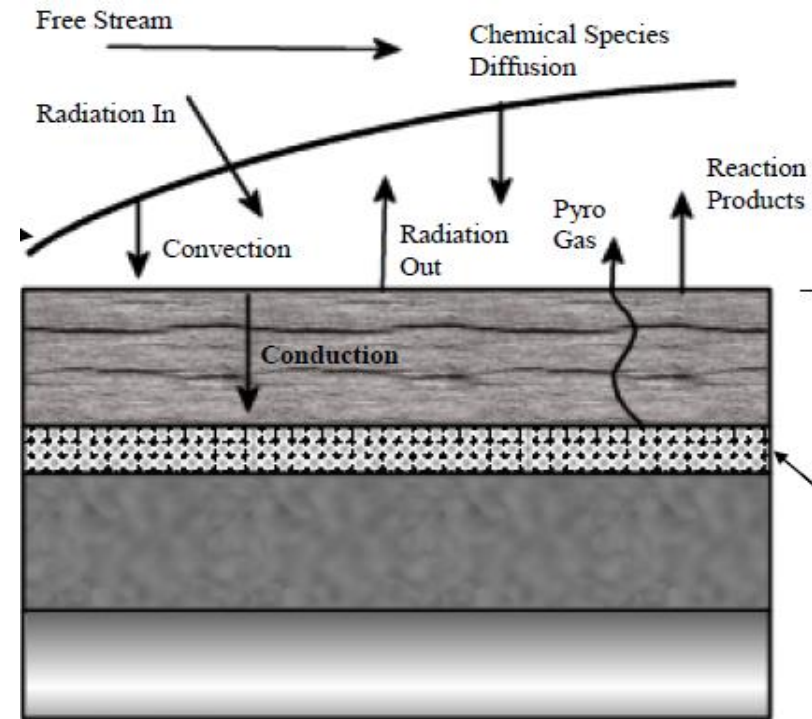
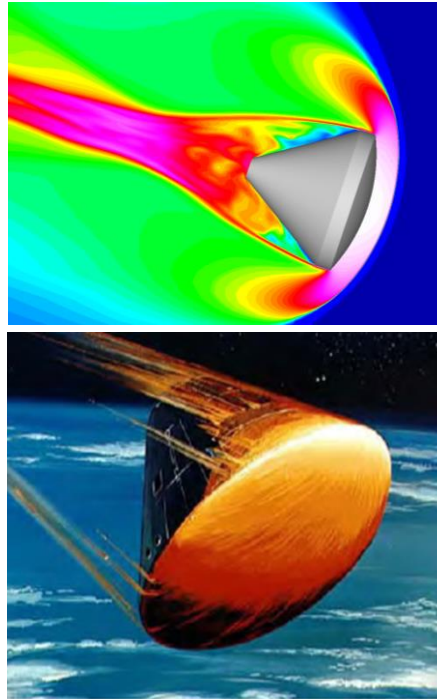
# Purpose of Task

Develop **ultrahigh temperature, light weight, low erosion, and cost effective** ablative thermal protection systems with embedded health monitoring for inherent safety and real-time assessment of TPS performance in hypersonic space vehicles

TPS for Ascent/Entry



Reentry  
Vehicles



# Purpose of Task

## RELEVANCE TO COMMERCIAL SPACE INDUSTRY

- Ultra-high temperature, light weight, low erosion, and cost effective thermal protection systems (TPS) are enabling technologies for viable commercial space transportation vehicles and their high-temperature systems.

## STATEMENT OF WORK

- Develop nanocomposites TPS with embedded health monitoring for inherent safety and real-time assessment of hypersonic TPS applications.
- Provide an analysis tool for the aerothermal modeling of reentry vehicles and rocket propulsion.
- Provide an analysis tool for thermal degradation modeling of new ablative materials.
- Provide ablation sensing to monitor the structural health of the ablative thermal protection system.

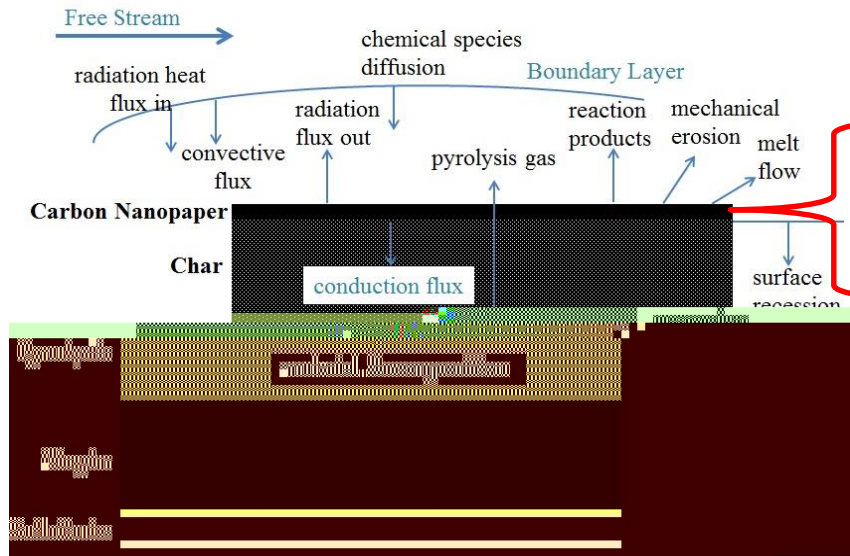
# Current Approach

- PICA: Phenolic Impregnated Carbon Ablator
- SICA: Silicone Impregnated Carbon Ablator
- Carbon/Carbon Composites

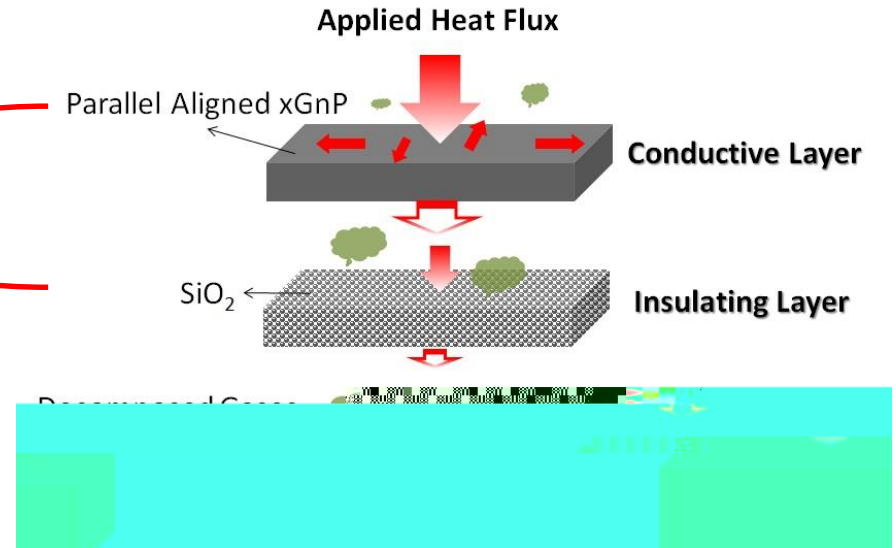
## Problems

- The resulting chars are structurally weak and susceptible to mechanical erosion, severely reducing the lifetime of the TPS. Reducing spallation or erosion of the char can enable use of less ablative materials thereby reducing the total weight of TPS.
- The evaluation of ablation performance needs to consider the structural integrity of TPS structures
- Recession monitoring is most important measurement to the aerothermal analysis of the TPS structure. This measurement provides critical information about how the TPS mass and shape changes during the flight.

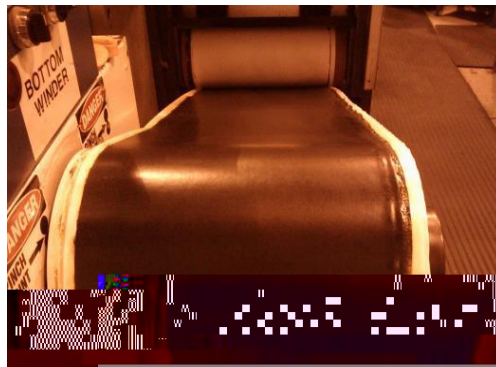
# Ablative TPS Design - Nanocomposites Approach



TPS Materials Design



Bi-layer thermal protective coating



Nanocomposite prepreg systems

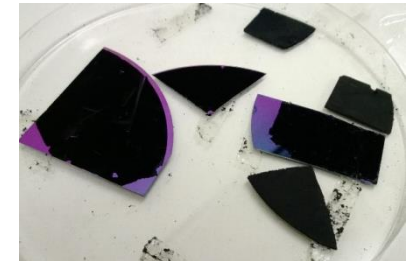
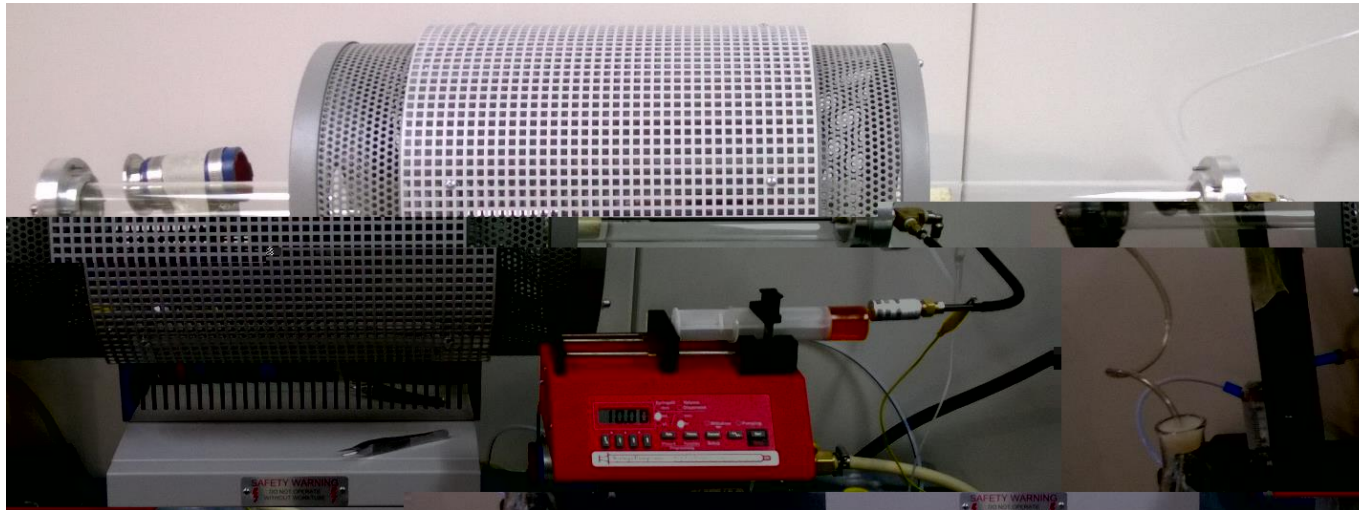


Nanocomposite thermal protective coating

# Research Methodology

## Vertically Aligned Carbon Nanotubes (VACNT)

- Aligned CNT Arrays grown by means of Chemical Vapor Deposition (CVD)
- Highly anisotropic properties
- Improved thermal soak rate
- Combinable with other technologies being explored





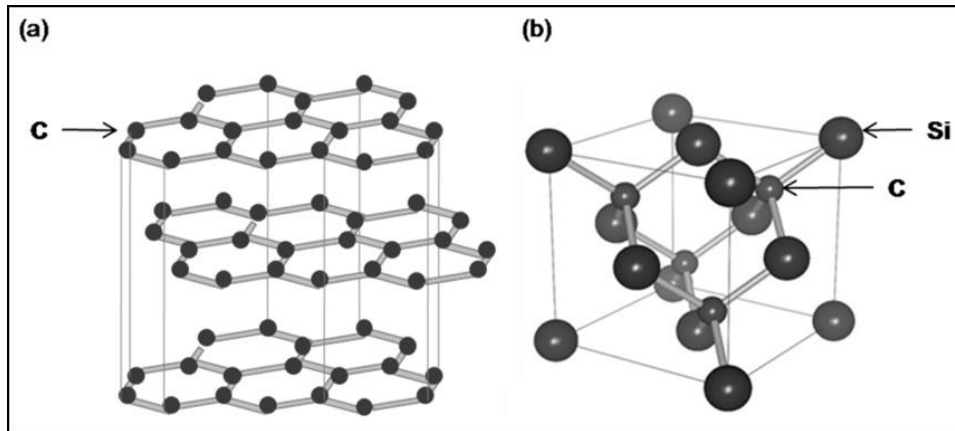
# Research Methodology

## Polymer Derived Ceramics (PDC)

- Corrosion and wear resistant
- Simple processing with moldable geometries
- Strong and high quality ceramic matrix yield
- Lower thermal material requirement

## Fiber Reinforced Ceramic Matrix Composites (CMC)

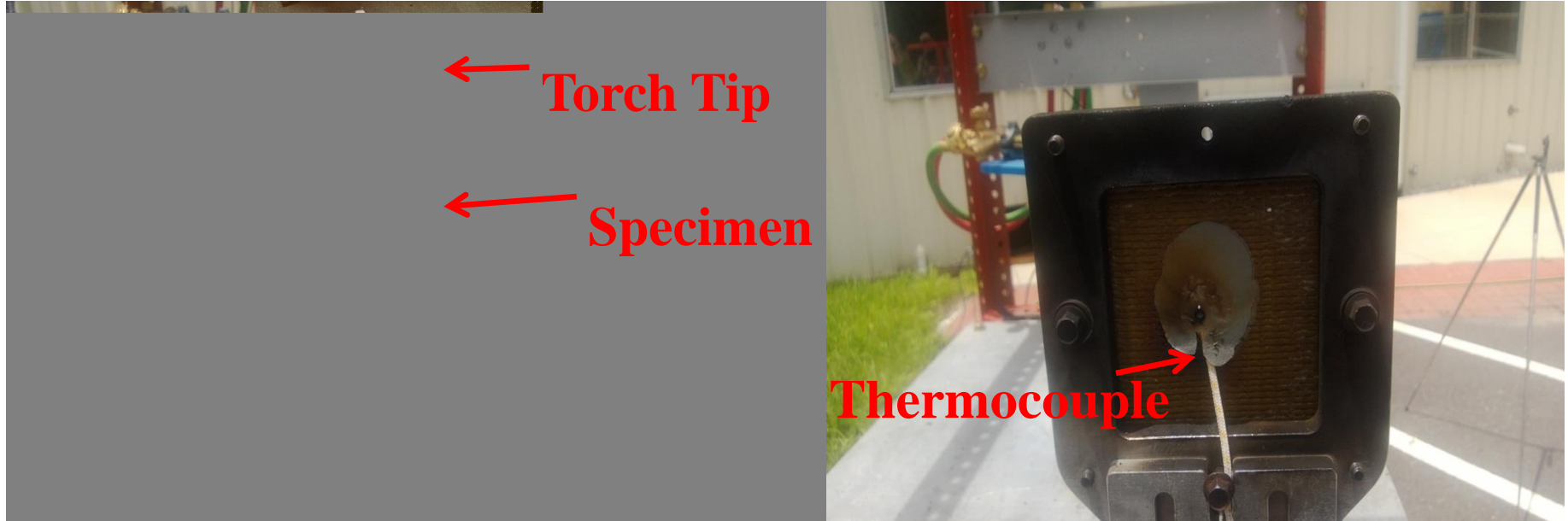
- High fracture toughness, crack resistance
- Resistance to thermal shock
- Higher oxidation temperatures ( $>1000^{\circ}\text{C}$ )



# Results

## PICA - Nanocomposite Thermal Protective Coating

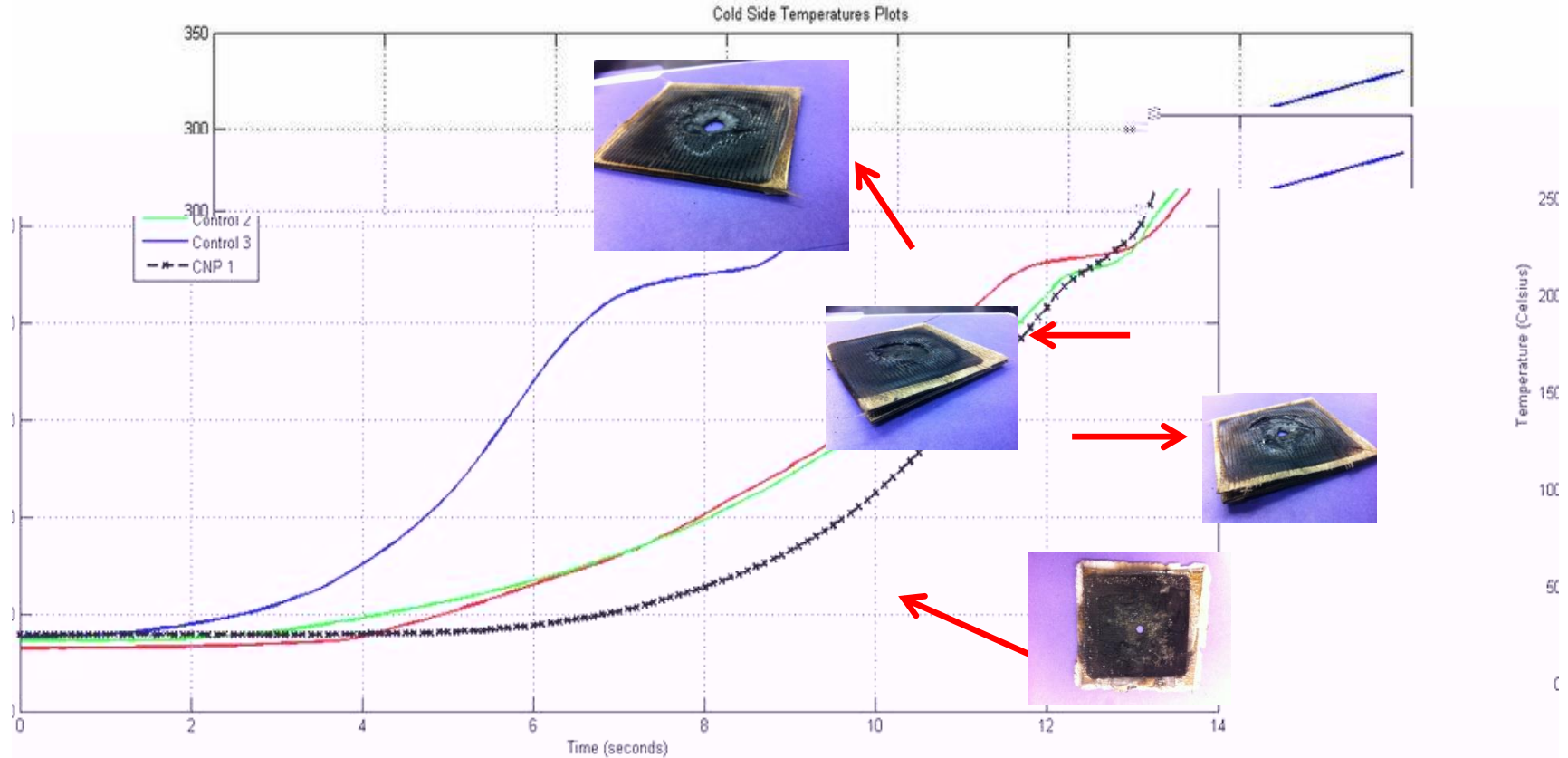
- Introduction of organized, structured carbon layer
- Improvement in insulation as well as ablative regression
- Delays first layer delamination



### Oxyacetylene Ablation Testing

# Results

## Cold Side Temperature from Ablation Test



A delay of the increase of cold side temperature for the specimen coated with nanopaper indicates the nanopaper provided good thermal protection for the composites at high heat flux.

# Next Steps

- Continue oxyacetylene torch testing of composite panels
- Integration of nanoparticles into PDC composites
- Incorporation of VACNT into composite panels
- Aerothermal analysis of TPS structures with curved geometry
- Thermomechanical characterization and testing for structural integrity evaluation

# Contact Information

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